# Combined Loading 

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## Contents

- Unsymmetric Bending（不对称弯曲）
- Tension \＆Bending（拉弯组合）
- Eccentric Compression（偏心压缩）
- Core of Cross－sections（截面核心区域）
- Core of Rectangular Cross－sections（矩形截面核心区域）
- Core of Circular Cross－sections（圆形截面核心区域）
- Tension \＆Torsion（拉扭组合）
- Bending \＆Torsion（弯扭组合）
- Tension，Bending \＆Torsion（拉弯扭组合）


## Introduction

- A circular bar subjected to a single type of load

|  | Stresses Produced by Each Load Individually | Stress <br> Distributions | Stresses |
| :---: | :---: | :---: | :---: |
| Torsional <br> Load <br> (Torque T) |  |  | Torsional shear $\begin{gathered} \text { stress } \\ \tau_{x \theta}=T \rho / I_{p} \end{gathered}$ |
| $\begin{gathered} \text { Axial } \\ \text { Load } \\ \text { (Force } F \text { ) } \end{gathered}$ |  |  | Tensile average normal stress $\sigma_{\text {avg }}=F / A$ |
| Bending Load (Transverse Force $P$ ) |  |  | Bending normal stress $\sigma_{x}=M_{z} y / I_{z}$ <br> Transverse shear stress $\tau_{x y}=F_{\mathrm{s}} S_{z}^{*} / I_{z} b$ |

## Introduction

- Prismatic bars are frequently subjected to several loads simultaneously
- The principle of superposition is used to determine the resultant stress \& strain
- Conditions for the principle of superposition
- Linear elasticity \& small deformation
- No interaction between variously loads



## Unsymmetric Bending



- Analysis of pure bending has been limited to members subjected to bending moments acting in a plane of symmetry.
- Members remain symmetric and bend in the plane of symmetry.
- The neutral axis of the cross section coincides with the axis of the couple
- Will now consider situations in which the bending couples do not act in a plane of symmetry.
- Cannot assume that the member will bend in the plane of the couples.
- In general, the neutral axis of the section will not coincide with the axis of the couple.


## Unsymmetric Bending



- If neutral axis passes through centroid

$$
\begin{aligned}
& 0=F_{x}=\int \sigma_{x} d A=\int \frac{M_{z} y}{I_{z}} d A \\
& \text { or } 0=\int y d A
\end{aligned}
$$

- Stress distribution

Wish to determine the conditions under which the neutral axis of a cross section of arbitrary shape coincides with the axis of the moment as shown.

- The resultant force and moment from the distribution of elementary forces in the section must satisfy
$F_{x}=0=M_{y} \quad M_{z}=M=$ applied couple

$$
M=M_{z}=\int \frac{M_{z} y}{I_{z}}(y) d A
$$

- Moment vector must be directed along a principal centroidal axis

$$
0=M_{y}=\int \frac{M_{z} y}{I_{z}}(z) d A
$$

or $\quad 0=\int y z d A=I_{y z}=$ product of inertia

- Superposition is applied to determine stresses in the most general case of unsymmetric bending.


## Unsymmetric Bending

- Sign Convention

- Construct a coordinate system.
- In $\boldsymbol{x} \boldsymbol{- y}$ plane: positive $M_{z}$ results in compression for $\boldsymbol{y}<0$.
- Bending stress and deflection: $\sigma_{x}=M_{z} y / I_{z}, w_{y}^{\prime \prime}=-M_{z} / E I_{z}$.
- The same sign conventions can be used for bending in $x-z$ plane.
- Positive $M_{y}$ results in compression for $z<0$.
- Bending stress and deflection: $\sigma_{x}=M_{y} z / I_{y}, w_{z}^{\prime \prime}=-M_{y} / E I_{y}$.


## Unsymmetric Bending


$\left.\begin{array}{l}M_{z}=F_{y}(x-a), \sigma_{x}^{\prime}=\frac{M_{z} y}{I_{z}}, \quad a \leq x \leq L \\ M_{y}=F_{z} x, \quad \sigma_{x}^{\prime \prime}=\frac{M_{y} z}{I_{y}}, \quad 0 \leq x \leq L\end{array}\right\} \Rightarrow \sigma_{x}=\sigma_{x}^{\prime}+\sigma_{x}^{\prime \prime}=\frac{M_{z} y}{I_{z}}+\frac{M_{y} z}{I_{y}}$

## Unsymmetric Bending



- Equation for Neutral axis:

$$
\begin{aligned}
& 0=\sigma_{x}=\frac{M_{z} y_{0}}{I_{z}}+\frac{M_{y} z_{0}}{I_{y}} \\
& \Rightarrow \tan \theta=\frac{-z_{0}}{y_{0}}=\frac{M_{z}}{M_{y}} \frac{I_{y}}{I_{z}}=\frac{F_{y}}{F_{z}} \frac{(x-a)}{x} \frac{I_{y}}{I_{z}} \\
& \quad=\tan \alpha \frac{I_{y}}{I_{z}}=\tan \beta \frac{(x-a)}{x} \frac{I_{y}}{I_{z}} \\
& I_{y} \neq I_{z} \Rightarrow \theta \neq \alpha \quad a \neq 0 \Rightarrow \alpha \neq \beta
\end{aligned}
$$

- The neutral axis passes through the centroid of the cross-section.
- The maximum stresses occur at the two farthest points from the neutral axis.
- With the exception of $I_{y}=I_{z}$, i.e. for circular/square cross-section, the bending stress and deflection cannot be calculated from the resultant moment.
- Bending in a single plane occurs if and only if the orientation of neural axis stays the same for every cross-section, i.e. $M_{z} / M_{y}=$ constant for a prismatic beam.
- In general, superposition should be resorted to determine both the bending stress and the deflection.


## Sample Problem



A 1600 lb -in couple is applied to a rectangular wooden beam in a plane forming an angle of 30 deg. with the vertical. Determine (a) the maximum stress in the beam, (b) the angle that the neutral axis forms with the horizontal plane.

## SOLUTION:

- Resolve the couple vector into components along the principle centroidal axes and calculate the corresponding maximum stresses.
- Combine the stresses from the component stress distributions.
- Determine the angle of the neutral axis.

- Resolve the couple vector into components and calculate the corresponding maximum stresses.

$$
\begin{aligned}
& M_{z}=(1600 \mathrm{lb} \cdot \mathrm{in}) \cos 30=1386 \mathrm{lb} \cdot \mathrm{in} \\
& M_{y}=-(1600 \mathrm{lb} \cdot \mathrm{in}) \sin 30=-800 \mathrm{lb} \cdot \mathrm{in} \\
& I_{z}=\frac{1}{12}(1.5 \mathrm{in})(3.5 \mathrm{in})^{3}=5.359 \mathrm{in}^{4} \\
& I_{y}=\frac{1}{12}(3.5 \mathrm{in})(1.5 \mathrm{in})^{3}=0.9844 \mathrm{in}^{4}
\end{aligned}
$$

The largest tensile stress due to $M_{z}$ occurs along $A B$

$$
\sigma_{1}=\frac{M_{z} y}{I_{z}}=\frac{(1386 \mathrm{lb} \cdot \mathrm{in})(1.75 \mathrm{in})}{5.359 \mathrm{in}^{4}}=452.6 \mathrm{psi}
$$

The largest tensile stress due to $M_{z}$ occurs along $A D$

$$
\sigma_{2}=\frac{M_{y} z}{I_{y}}=\frac{(-800 \mathrm{lb} \cdot \mathrm{in})(-0.75 \mathrm{in})}{0.9844 \mathrm{in}^{4}}=609.5 \mathrm{psi}
$$

- The largest tensile stress due to the combined loading occurs at $A$.

$$
\sigma_{\max }=\sigma_{1}+\sigma_{2}=452.6+609.5=1062 \mathrm{psi}
$$

- Determine the angle of the neutral axis.

$$
0=\sigma_{x}=\frac{M_{z} y_{0}}{I_{z}}+\frac{M_{y} z_{0}}{I_{y}}=\frac{(1386 \mathrm{lb} \cdot \mathrm{in}) y_{0}}{5.359 \mathrm{in}^{4}}+\frac{(-800 \mathrm{lb} \cdot \mathrm{in}) z_{0}}{0.9844}=258.63 y_{0}-812.68 z_{0}
$$

$\tan \phi=\frac{y_{0}}{z_{0}}=\frac{812.68}{258.63}=3.142$

$$
\phi=72.4^{\circ}
$$



## Sample Problem

- For the $I-32 a$ beam shown, $L=4 \mathrm{~m},[\sigma]=160 \mathrm{MPa}, F=80 \mathrm{kN}, \alpha=$ $5^{\circ}$. Analyze the strength condition.
- Solution


1. Decompose the load
2. Critical cross-section

$$
x=\frac{L}{2} ; M_{z}=\frac{F_{y} L}{4} ; \quad M_{y}=-\frac{F_{z} L}{4}
$$

3. Neutral axis \& maximum stresses

$$
\sigma_{x}=\frac{M_{y} z}{I_{y}}+\frac{M_{z} y}{I_{z}}=-\frac{F_{z} L}{4 I_{y}} z+\frac{F_{y} L}{4 I_{z}} y
$$


4. Strength check:

$$
\sigma_{\max }=\frac{M_{y}}{W_{y}}+\frac{M_{z}}{W_{z}}=217.8 \mathrm{MPa}>[\sigma]=160 \mathrm{MPa}
$$

5. Deflections: $w=\sqrt{w_{y}^{2}+w_{z}^{2}}$
6. If $\alpha=0, \sigma_{\max }=\frac{M_{z}}{W_{z}}=115.6 M P a<[\sigma]$


Remark: The fact that the neutral axis is not necessarily perpendicular to the resultant moment can greatly affect the stresses in a beam, especially if the ratio of the principal moments of inertia is very large. Under these conditions the stresses in the beam are very sensitive to slight changes in the direction of the load and to irregularities in the alignment of the beam itself.

## Tension \& Bending (Eccentric Tension)



- Stress due to eccentric loading found by superposing the uniform stress due to a centric load and linear stress distribution due a pure bending moment

$$
\begin{aligned}
\sigma_{x} & =\left(\sigma_{x}\right)_{\text {centric }}+\left(\sigma_{x}\right)_{\text {bending }} \\
& =\frac{P}{A}+\frac{M_{z} y}{I_{z}}
\end{aligned}
$$

- Strength condition:

$$
\sigma_{\max }^{+} \leq\left[\sigma^{+}\right],\left|\sigma_{\max }^{-}\right| \leq\left[\sigma^{-}\right] .
$$



- Validity requires stresses below proportional limit, deformations have negligible effect on geometry, and stresses not evaluated near points of load application.



## Sample Problem



An open-link chain is obtained by bending low-carbon steel rods into the shape shown. For 160 lb load, determine (a) maximum tensile and compressive stresses, (b) distance between section centroid and neutral axis

## SOLUTION:

- Find the equivalent centric load and bending moment
- Superpose the uniform stress due to the centric load and the linear stress due to the bending moment.
- Evaluate the maximum tensile and compressive stresses at the inner and outer edges, respectively, of the superposed stress distribution.
- Find the neutral axis by determining the location where the normal stress is zero.

- Equivalent centric load and bending moment

$$
\begin{aligned}
P & =160 \mathrm{lb} \\
M_{z} & =-P d \\
& =-(160 \mathrm{lb})(0.65 \mathrm{in}) \\
& =-104 \mathrm{lb} \cdot \mathrm{in}
\end{aligned}
$$

- Normal stress due to a centric load

$$
\begin{aligned}
A & =\pi c^{2}=\pi(0.25 \mathrm{in})^{2} \\
& =0.1963 \mathrm{in}^{2} \\
\sigma_{x} & =\frac{P}{A}=\frac{160 \mathrm{lb}}{0.1963 \mathrm{in}^{2}} \\
& =815 \mathrm{psi}
\end{aligned}
$$

- Normal stress due to bending moment

$$
\begin{aligned}
I_{z} & =\frac{1}{4} \pi c^{4}=\frac{1}{4} \pi(0.25)^{4} \\
& =3.068 \times 10^{-3} \mathrm{in}^{4} \\
\sigma_{x} & =\frac{M_{z} c}{I_{z}}=\frac{(104 \mathrm{lb} \cdot \mathrm{in})(0.25 \mathrm{in})}{3.068 \times 10^{-3} \mathrm{in}^{4}} \\
& =8475 \mathrm{psi}
\end{aligned}
$$



- Maximum tensile and compressive stresses

$$
\begin{aligned}
& \sigma_{\max }^{+}=815+8475=9260 \mathrm{psi} \\
& \sigma_{\max }^{-}=815-8475=-7660 \mathrm{psi}
\end{aligned}
$$

- Neutral axis

$$
\begin{aligned}
0 & =\frac{P}{A}+\frac{M_{z} y_{0}}{I_{z}} \\
y_{0} & =-\frac{P}{A} \frac{I_{z}}{M_{z}}=-\frac{160 \mathrm{lb}}{0.1963 \mathrm{in}^{2}} \frac{3.068 \times 10^{-3} \mathrm{in}^{4}}{(-104 \mathrm{lb} \cdot \mathrm{in})} \\
& =0.0240 \mathrm{in}
\end{aligned}
$$

## Eccentric Compression




- Bars subjected to axial forces deviated a distance from axis.

1. Find the equivalent system of forces at the centroid.
2. Normal stress on cross-sections.

$$
\begin{aligned}
& F_{\mathrm{N}}=-F, \sigma_{x}^{\prime}=\frac{F_{N}}{A}=-\frac{F}{A} \\
& M_{z}=-F \cdot y_{F}, \sigma_{x}^{\prime \prime}=\frac{M_{z} \cdot y}{I_{z}}=-\frac{F \cdot y_{F} \cdot y}{I_{z}} \\
& M_{y}=-F \cdot z_{F}, \sigma_{x}^{\prime \prime \prime}=\frac{M_{y} \cdot z}{I_{y}}=-\frac{F \cdot z_{F} \cdot z}{I_{y}} \\
& \Rightarrow \sigma=\sigma^{\prime}+\sigma^{\prime \prime}+\sigma^{\prime \prime \prime} \\
&=-\left(\frac{F}{A}+\frac{F \cdot y_{F} \cdot y}{I_{z}}+\frac{F \cdot z_{F} \cdot z}{I_{y}}\right) \\
& I_{y}=A i_{y}^{2}, I_{z}=A i_{z}^{2} \Rightarrow \sigma=-\frac{F}{A}\left(1+\frac{y_{F} y}{i_{z}^{2}}+\frac{z_{F} z}{i_{y}^{2}}\right)
\end{aligned}
$$



- Note the negative sign in bending stresses.


3. Equation of neutral axis.

$$
\begin{aligned}
& 0=\sigma=-\frac{F}{A}\left(1+\frac{y_{F} y}{i_{z}^{2}}+\frac{z_{F} z}{i_{y}^{2}}\right) \\
& \Rightarrow 1+\frac{y_{F} y_{0}}{i_{z}^{2}}+\frac{z_{F} z_{0}}{i_{y}^{2}}=0
\end{aligned}
$$

- The intercepts of neutral axis:


$$
a_{y}=\left.y_{0}\right|_{z_{0}=0}=-\frac{i_{z}^{2}}{y_{F}} ; \quad \boldsymbol{a}_{z}=\left.z_{0}\right|_{y_{0}=0}=-\frac{\boldsymbol{i}_{y}^{2}}{z_{F}}
$$

4. Strength condition: $\sigma_{\max }^{+} \leq\left[\sigma^{+}\right],\left|\sigma_{\max }^{-}\right| \leq\left[\sigma^{-}\right]$.

## Sample Problem



The largest allowable stresses for the cast iron link are 30 MPa in tension and 120 MPa in compression. Determine the largest force $P$ which can be applied to the link and the neutral axis.

## SOLUTION:

- Determine an equivalent centric load and bending moment.
- Superpose the stress due to a centric load and the stress due to bending.
- Evaluate the critical loads for the allowable tensile and compressive stresses.
- The largest allowable load is the smallest of the two critical loads.

From geometry,


- Determine an equivalent centric and bending loads.

$$
\begin{aligned}
& d=0.038-0.010=0.028 \mathrm{~m} \\
& M_{z}=-P d=-0.028 P=\text { bending moment }
\end{aligned}
$$

- Superpose stresses due to centric and bending loads

$$
\begin{aligned}
& \sigma_{A}=-\frac{P}{A}+\frac{M_{z} c_{A}}{I_{z}}=-\frac{P}{3 \times 10^{-3}}+\frac{(-0.028 P)(-0.022)}{868 \times 10^{-9}}=+377 P \\
& \sigma_{B}=-\frac{P}{A}+\frac{M_{z} c_{B}}{I_{z}}=-\frac{P}{3 \times 10^{-3}}+\frac{(-0.028 P)(0.038)}{868 \times 10^{-9}}=-1559 P
\end{aligned}
$$

- Evaluate critical loads for allowable stresses.

$$
\begin{array}{ll}
\sigma_{A}=+377 P=30 \mathrm{MPa} & P=79.6 \mathrm{kN} \\
\sigma_{B}=-1559 P=-120 \mathrm{MPa} & P=77.0 \mathrm{kN}
\end{array}
$$

- The largest allowable load $\quad P=77.0 \mathrm{kN}$
- Neutral axis

$$
\begin{aligned}
& a_{y}=\left.y_{0}\right|_{z_{0}=0}=-\frac{i_{z}^{2}}{y_{F}}=-\frac{868 \times 10^{-9} / 3 \times 10^{-3}}{0.028}=-10.33 \mathrm{~mm} \\
& a_{z}=\left.z_{0}\right|_{y_{0}=0}=-\frac{i_{y}^{2}}{z_{F}} \rightarrow \infty
\end{aligned}
$$

## Sample Problem

- Find the maximum normal stress for the bar shown.
- Solution:

1. Equivalent System of forces

$$
\begin{aligned}
& F_{\mathrm{N}}=F=10 \mathrm{kN} \\
& M_{z}=10 \times 10^{3} \times 5 \times 10^{-2}=500 \mathrm{Nm} \\
& M_{y}=10 \times 10^{3} \times 2.5 \times 10^{-2}=250 \mathrm{Nm}
\end{aligned}
$$

2. Maximum normal stress

$$
\begin{aligned}
& \sigma_{\text {max }}^{-}=\sigma_{A}=-\frac{F_{N}}{A}+\frac{M_{z} y_{A}}{I_{z}}+\frac{M_{y} z_{A}}{I_{y}}=-14 \mathrm{MPa} \\
& \sigma_{\text {max }}^{+}=\sigma_{B}=-\frac{\boldsymbol{F}_{N}}{A}+\frac{\boldsymbol{M}_{z} y_{B}}{\boldsymbol{I}_{z}}+\frac{\boldsymbol{M}_{y} z_{B}}{\boldsymbol{I}_{y}}=10 \mathrm{MPa}
\end{aligned}
$$

Intercepts: $\boldsymbol{a}_{y}=-\frac{\boldsymbol{i}_{z}^{2}}{\boldsymbol{y}_{F}} ; \quad \boldsymbol{a}_{z}=-\frac{\boldsymbol{i}_{y}^{2}}{z_{F}}$


View from left

## Core of Cross-sections

- Portion of a cross-section within which eccentric loading results in only compressive stresses.
- Intercepts of neutral axis:

$$
\begin{aligned}
& a_{y}=-\frac{i_{z}^{2}}{y_{F}} ; a_{z}=-\frac{i_{y}^{2}}{z_{F}} \\
& \Rightarrow y_{F}=-\frac{i_{z}^{2}}{a_{y}} ; z_{F}=-\frac{i_{y}^{2}}{a_{z}}
\end{aligned}
$$



## Core of Rectangular Cross-sections

$$
\begin{array}{ll}
y_{F}=-\frac{i_{z}^{2}}{a_{y}}, & z_{F}=-\frac{\boldsymbol{i}_{y}^{2}}{\boldsymbol{a}_{z}} \\
\boldsymbol{i}_{z}^{2}=\frac{\boldsymbol{I}_{z}}{\boldsymbol{A}}=\frac{\boldsymbol{h}^{2}}{12}, & i_{y}^{2}=\frac{I_{y}}{A}=\frac{b^{2}}{12}
\end{array}
$$

- Take side $A B$ as neutral axis


$$
\begin{gathered}
\boldsymbol{a}_{\boldsymbol{y}_{1}}=\infty, \quad \boldsymbol{a}_{z_{1}}=-\frac{\boldsymbol{b}}{2} \\
\Rightarrow y_{F_{1}}=-\frac{i_{z}^{2}}{a_{y_{1}}}=0, \quad z_{F_{1}}=-\frac{i_{y}^{2}}{a_{z_{1}}}=\frac{b}{6}
\end{gathered}
$$

- Take side $B C$ as neutral axis

$$
y_{F_{2}}=-\frac{\boldsymbol{h}}{6}, \quad z_{F_{2}}=0
$$

- Through corner $B$, there exists infinite number of neutral axes we can take.

$$
\begin{aligned}
& 1+\frac{y_{F} y_{B}}{i_{z}^{2}}+\frac{z_{F} z_{B}}{i_{y}^{2}}=0 \\
& \Rightarrow 1+\frac{y_{F}}{h^{2} / 12} h / 2+\frac{z_{F}}{b^{2} / 12}(-b / 2)=0 \\
& \Rightarrow 1+\frac{6 y_{F}}{h}-\frac{6 z_{F}}{b}=0
\end{aligned}
$$

## Core of Circular Cross-sections



- Due to axial symmetry, only one neutral axis is necessary to determine the radius of the core area:

$$
\begin{aligned}
& \text { (1): } a_{y}=\infty, a_{z}=d / 2 \\
& \quad i_{y}=i_{z}=d / 4 \\
& \quad \Rightarrow y_{F 1}=-\frac{i_{z}^{2}}{a_{y}}=0, \quad z_{F 1}=-\frac{i_{y}^{2}}{a_{z}}=-\frac{(d / 4)^{2}}{d / 2}=-d / 8
\end{aligned}
$$

- Note: no matter whatever shape of a cross-section area, the corresponding core area is always solid and non-concave.



## Tension \& Torsion



- Solution


$$
\begin{aligned}
& \sigma_{x}=\frac{F_{N}}{A}=\frac{4 P}{\pi d^{2}} \\
& \tau_{x \theta}=\frac{T}{W_{P}}=\frac{T}{\pi d^{3} / 16}
\end{aligned}
$$

## Bending \& Torsion



- Critical section: $A$


View from left View from right

- Critical points on $A: A_{1} \& A_{2}$

$$
\sigma=\frac{\boldsymbol{M}_{z}}{\boldsymbol{W}_{z}}, \quad \tau=\frac{\boldsymbol{T}}{\boldsymbol{W}_{\boldsymbol{P}}}
$$

$$
\begin{aligned}
& \sigma_{r 3} \leq[\sigma] \\
& \sigma_{r 4} \leq[\sigma]
\end{aligned} \quad \begin{aligned}
& \sqrt{\sigma^{2}+4 \tau^{2}} \leq[\sigma] \\
& \sqrt{\sigma^{2}+3 \tau^{2}} \leq[\sigma]
\end{aligned}
$$

$$
\Rightarrow\left\{\begin{array}{l}
\sigma_{r 3}=\sqrt{\left(\frac{M_{z}}{W_{z}}\right)^{2}+4\left(\frac{T}{W_{P}}\right)^{2}}=\frac{1}{W_{z}} \sqrt{M_{z}^{2}+T^{2}} \leq[\sigma]  \tag{29}\\
\sigma_{r 4}=\sqrt{\left(\frac{M_{z}}{W_{z}}\right)^{2}+3\left(\frac{T}{W_{P}}\right)^{2}}=\frac{1}{W_{z}} \sqrt{M_{z}^{2}+0.75 T^{2}} \leq[\sigma]
\end{array}\right.
$$

## Sample Problem

- For the circular shaft shown, $L=50 \mathrm{~cm}, F=8 \mathrm{kN}, a=37.5 \mathrm{~cm},[\sigma]=$ 100 MPa . Find the minimum diameter of shaft $A B$ based on maximum shearing stress criteria.

- Solution:

$$
T=3 \mathrm{kNm}, \quad M_{\max }=F L=4 \mathrm{kNm}
$$



$$
30
$$

## Tension, Bending \& Torsion - an Exercise

- Two forces $P=18 \mathrm{kN}$ and $F=15 \mathrm{kN}$ are applied to the shaft with a radius of $R=20 \mathrm{~mm}$ as shown. Determine the maximum normal and shearing stresses developed in the shaft.



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